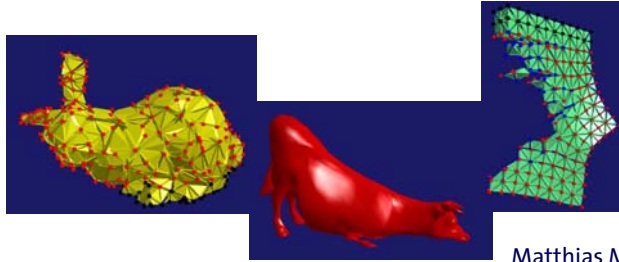


Interactive Simulation of Elasto-Plastic Materials using the Finite Element Method



Matthias Müller
Seminar – Wintersemester 02/03

Outline

- FEM vs. Mass-Spring**
- Stiffness**
 - The Stiffness Matrix
 - Static/Dynamic Deformation
- Continuum Mechanics and FEM**
 - Strain and Stress Tensors
 - Continuous PDE's
 - FEM Discretization
- Plasticity**
 - Plastic Strain
 - Update Rules
- Fracture**
 - Principal Stresses
 - Crack Computation

Mass-Spring vs. FEM

- | | |
|---|--|
| <ol style="list-style-type: none"> 1. Discretization of an object into mass points 2. Representation of forces between mass points with springs 3. Computation of the dynamics | <ol style="list-style-type: none"> 1. Discretization of an object into elements (tetrahedra) 2. Discretization of continuous energy equations into algebraic equations for forces acting at vertices 3. Computation of the dynamics |
|---|--|

➡ deformable mass-spring system

➡ deformable FEM system

Pros and Cons of FEM

- Pros:
- No individual spring constants needed (only 2 known material parameters E, ν)
 - No inversion problems (inverted tetrahedra produce forces)
 - Stress and strain tensors allow
 - fracture and
 - plasticity simulations
- Cons:
- (Pre-)compute stiffness matrix
 - Store stiffness matrix (3×3) per edge
 - Store original **and** actual positions of vertices



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Continuous PDE's
FEM Discretization

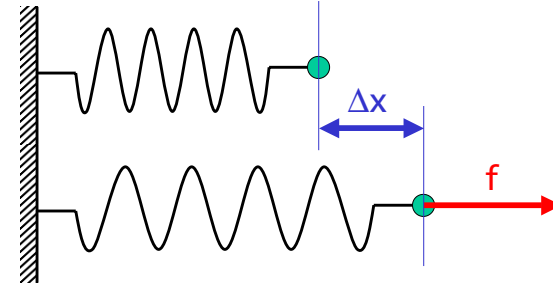
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One-dimensional Spring



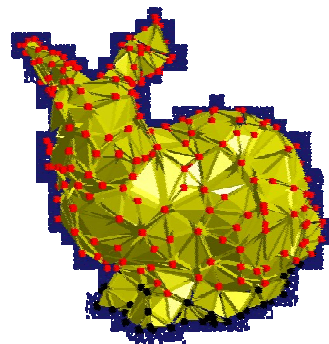
$$f = k \cdot \Delta x$$

Three-dimensional Object

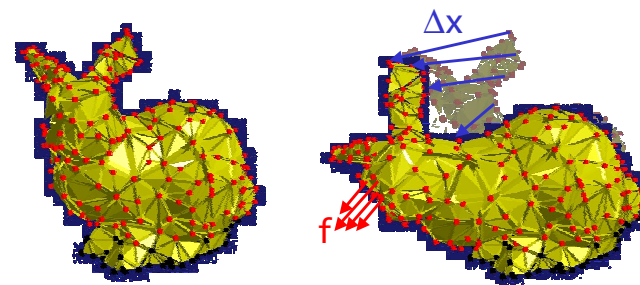
Finite Element Mesh

- 903 tetrahedra
- 393 vertices
- $3 \times 393 = 1179$ dof.

$$\Delta \mathbf{x} = \begin{bmatrix} \Delta x_{1x} \\ \Delta x_{1y} \\ \Delta x_{1z} \\ \dots \\ \Delta x_{nx} \\ \Delta x_{ny} \\ \Delta x_{nz} \end{bmatrix} \quad \mathbf{f} = \begin{bmatrix} f_{1x} \\ f_{1y} \\ f_{1z} \\ \dots \\ f_n \\ f_{ny} \\ f_{nz} \end{bmatrix} \times$$



Three-dimensional Object



$$\mathbf{f}_{el} = \mathbf{K} \cdot \Delta \mathbf{x} \quad (\text{Stiffness Matrix } \mathbf{K} \in \mathbb{R}^{3n \times 3n})$$

$$\mathbf{f}_{el} = \mathbf{F}(\Delta \mathbf{x}) \quad (\text{Function } \mathbf{F} : \mathbb{R}^{3n} \rightarrow \mathbb{R}^{3n})$$

Static Deformation

$$\mathbf{f}_{\text{ext}} = \mathbf{f}_{\text{el}} = \mathbf{K} \cdot \Delta \mathbf{x}$$

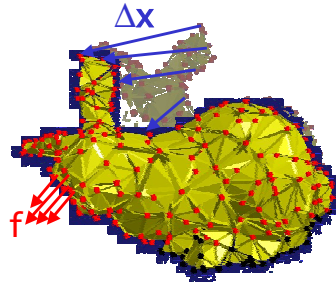
$$\Delta \mathbf{x} = \mathbf{K}^{-1} \cdot \mathbf{f}_{\text{ext}}$$

Solve linear system
(Conjugate Gradients)

$$\mathbf{f}_{\text{ext}} = \mathbf{f}_{\text{el}} = \mathbf{F}(\Delta \mathbf{x})$$

$$\Delta \mathbf{x} = \mathbf{F}^{-1}(\mathbf{f}_{\text{ext}})$$

Solve non-linear system
(Newton-Raphson - generalized Newton-Method)



Dynamic Deformation

$$\mathbf{M}\mathbf{x}'' + \mathbf{C}\mathbf{x}' + \mathbf{K}\Delta \mathbf{x} = \mathbf{f}_{\text{ext}}$$

- Coupled system of 3n linear ODEs
- Explicit integration: No solver needed
- Implicit integration: Linear solver per time step

$$\mathbf{M}\mathbf{x}'' + \mathbf{C}\mathbf{x}' + \mathbf{F}(\Delta \mathbf{x}) = \mathbf{f}_{\text{ext}}$$

- Coupled system of 3n non-linear ODEs
- Explicit integration: No solver needed
- Implicit integration: Linearize at every time step: $\mathbf{K} = d\mathbf{F}/d\mathbf{x}$

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Continuous PDE's
FEM Discretization

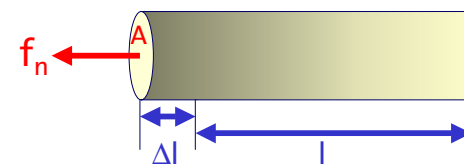
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Continuous Elasticity 1-d



stress σ [N/m²]
(Normal-Spannung)

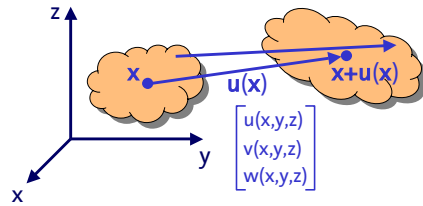
$$\mathbf{f}_n / A = E \Delta l / l$$

strain ϵ [1]
(Dehnung)

Elasticity (Young's) Modulus
[N/m²]
Metal: $\sim 10^{11}$ N/m²
Soft material: $\sim 10^6$ N/m²

Continuous Elasticity 3-d

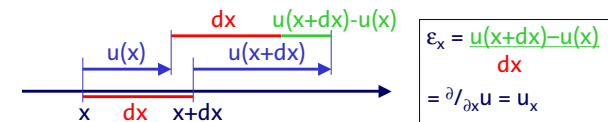
Deformation:



- Continuous 3-d vector field $u: \mathbb{R}^3 \rightarrow \mathbb{R}^3$
- Defined within undeformed object

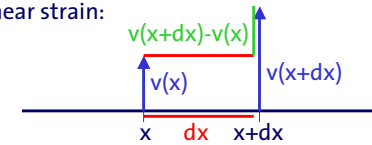
Linear 3-d Strain

normal strain in x - direction:



$$\epsilon_x = \frac{u(x+dx) - u(x)}{dx} = \partial / \partial x u = u_x$$

shear strain:



$$\gamma_{xy} = \frac{v(x+dx) - v(x)}{dx} = \partial / \partial x v = v_x$$

Linear 3-d Strain

Linear strain tensor:

$$\epsilon = \epsilon(x, y, z) = \begin{bmatrix} \epsilon_x & \gamma_{xy} & \gamma_{xz} \\ \gamma_{yx} & \epsilon_y & \gamma_{yz} \\ \gamma_{zx} & \gamma_{zy} & \epsilon_z \end{bmatrix} = \begin{bmatrix} u_x & u_y + v_x & u_z + w_x \\ v_x + u_y & v_y & v_z + w_y \\ w_x + u_z & w_y + v_z & w_z \end{bmatrix}$$

Symmetric, 3x3 matrix \rightarrow 6 vector:

$$\epsilon = \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \epsilon_z \\ \gamma_{xy} \\ \gamma_{yz} \\ \gamma_{zx} \end{bmatrix} = \begin{bmatrix} \partial / \partial x & 0 & 0 \\ 0 & \partial / \partial y & 0 \\ 0 & 0 & \partial / \partial z \\ \partial / \partial y & \partial / \partial x & 0 \\ 0 & \partial / \partial z & \partial / \partial y \\ \partial / \partial z & 0 & \partial / \partial x \end{bmatrix} \begin{bmatrix} u \\ v \\ w \end{bmatrix}$$

Non-Linear 3-d Strain

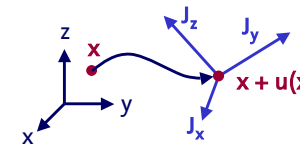
Green-Saint-Venant strain tensor:

- Transformation we use: $\mathbf{x} \rightarrow \mathbf{x} + \mathbf{u}(\mathbf{x})$:
- Use Jacobian of transformation:

$$\mathbf{J} = (\mathbf{J}_x, \mathbf{J}_y, \mathbf{J}_z) = \begin{pmatrix} \partial / \partial x(x+u) & \partial / \partial y(x+u) & \partial / \partial z(x+u) \\ \partial / \partial x(y+v) & \partial / \partial y(y+v) & \partial / \partial z(y+v) \\ \partial / \partial x(z+w) & \partial / \partial y(z+w) & \partial / \partial z(z+w) \end{pmatrix}$$

$$\epsilon_{\text{Green}} = \mathbf{J}^T \mathbf{J} - \mathbf{I}$$

- Interpretation:

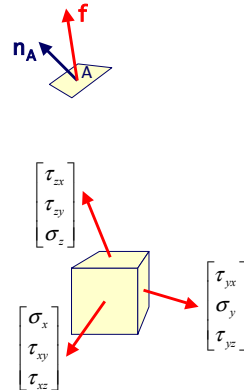


3-d Stress

Stress is force per (oriented) area:

$$\boldsymbol{\sigma} = \frac{d\mathbf{f}}{dA} = \frac{d\mathbf{f}}{dA \cdot \mathbf{n}_A}$$

$$\frac{d\mathbf{f}}{dA} = \boldsymbol{\sigma} \cdot \mathbf{n}_A = \begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{xy} & \sigma_y & \tau_{yz} \\ \tau_{xz} & \tau_{yz} & \sigma_z \end{bmatrix} \cdot \mathbf{n}_A$$



Constitutive Relation (isotropic)

The stress tensor is symmetric, 3x3 matrix → 6 vector:

Hooke's law: $\boldsymbol{\sigma} = \mathbf{E} \boldsymbol{\varepsilon}$

$$\begin{bmatrix} \sigma_x \\ \sigma_y \\ \sigma_z \\ \tau_{xy} \\ \tau_{yz} \\ \tau_{zx} \end{bmatrix} = \begin{bmatrix} (1-\nu)c & \nu c & \nu c & 0 & 0 & 0 \\ \nu c & (1-\nu)c & \nu c & 0 & 0 & 0 \\ \nu c & \nu c & (1-\nu)c & 0 & 0 & 0 \\ 0 & 0 & 0 & G & 0 & 0 \\ 0 & 0 & 0 & 0 & G & 0 \\ 0 & 0 & 0 & 0 & 0 & G \end{bmatrix} \begin{bmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_z \\ \gamma_{xy} \\ \gamma_{yz} \\ \gamma_{zx} \end{bmatrix}$$

$$c = \frac{E}{(1+\nu)(1-2\nu)}, G = \frac{E}{2(1+\nu)}$$

Only two scalar parameters:

E: Young's modulus, ν : Poisson ratio

Putting it all together

Given $\mathbf{u}(\mathbf{x})$ we can compute

- strain $\boldsymbol{\varepsilon}(\mathbf{x})$ and
- stress $\boldsymbol{\sigma}(\mathbf{x}) = \mathbf{E} \boldsymbol{\varepsilon}(\mathbf{x})$

at every point \mathbf{x} within the object.

→ Find $\mathbf{u}(\mathbf{x})$ such that corresponding stresses $\boldsymbol{\sigma}(\mathbf{x})$ are in balance with external forces $\mathbf{f}(\mathbf{x})$ everywhere within object:

$$\begin{bmatrix} \sigma_{x,x} + \tau_{xy,y} + \tau_{xz,z} + f_x = 0 \\ \tau_{yx,x} + \sigma_{y,y} + \tau_{yz,z} + f_y = 0 \\ \tau_{zx,x} + \tau_{zy,y} + \sigma_{z,z} + f_z = 0 \end{bmatrix}$$

- Strong formulation
- Coupled system of partial differential equations!

Energy Formulation

- Energy U is a scalar
- U at point \mathbf{x} is given by „displacement \times force“:

$$U_{\text{elastic}} = \frac{1}{2} \boldsymbol{\varepsilon}^T \boldsymbol{\sigma} = \frac{1}{2} \boldsymbol{\varepsilon}^T \mathbf{E} \boldsymbol{\varepsilon}$$

- The total Energy of the deformed body:

$$U_{\text{body}} = \int_{\text{body}} \left(\frac{1}{2} \boldsymbol{\varepsilon}^T \mathbf{E} \boldsymbol{\varepsilon} - \mathbf{u} \cdot \mathbf{f} \right) dV$$

- Given \mathbf{f} , \mathbf{E} we can compute U_{body} for any $\mathbf{u}(\mathbf{x})$
- Find $\mathbf{u}(\mathbf{x})$ such that U_{body} is a minimum ($\delta U = 0$)

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Plasticity

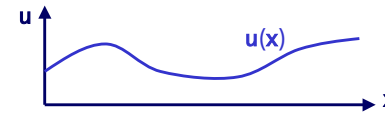
Plastic Strain
Update Rules

Fracture

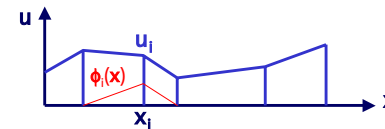
Principal Stresses
Crack Computation

Finite Element Formulation

- So far we looked for a continuous field $u(x)$

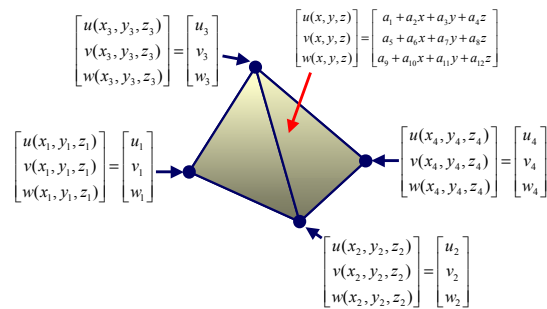


- Now we look for u_1, u_2, \dots, u_n at fixed locations: x_1, x_2, \dots, x_n
- and interpolate $u(x)$ with fixed basis functions: $u(x) \approx \sum_i u_i \phi_i(x)$



Linear Displacement Tetrahedron

- 12 unknowns (a_1, \dots, a_{12}), 12 equations
- u_i, v_i, w_i are variables, x_i, y_i, z_i are given numbers



Displacements

The displacement function $u(x)$ can be expressed as

- a matrix of basis functions $H(x)$ times
- a vector of displacements:

$$u(x) = H(x) \cdot \hat{u}$$

$$\begin{bmatrix} u(x, y, z) \\ v(x, y, z) \\ w(x, y, z) \end{bmatrix} = \begin{bmatrix} N_1 & 0 & 0 & N_2 & 0 & 0 & N_3 & 0 & 0 & N_4 & 0 & 0 \\ 0 & N_1 & 0 & 0 & N_2 & 0 & 0 & N_3 & 0 & 0 & N_4 & 0 \\ 0 & 0 & N_1 & 0 & 0 & N_2 & 0 & 0 & N_3 & 0 & 0 & N_4 \end{bmatrix} \cdot \begin{bmatrix} u_1 \\ v_1 \\ w_1 \\ u_2 \\ v_2 \\ w_2 \\ u_3 \\ v_3 \\ w_3 \\ u_4 \\ v_4 \\ w_4 \end{bmatrix}$$

Strain

Linear displacements yield constant strain:

$$\boldsymbol{\varepsilon} = \begin{bmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_z \\ \gamma_{xy} \\ \gamma_{yz} \\ \gamma_{zx} \end{bmatrix} = \begin{bmatrix} \partial/\partial x & 0 & 0 \\ 0 & \partial/\partial y & 0 \\ 0 & 0 & \partial/\partial z \\ \partial/\partial y & \partial/\partial x & 0 \\ 0 & \partial/\partial z & \partial/\partial y \\ \partial/\partial z & 0 & \partial/\partial x \end{bmatrix} \mathbf{H}(\mathbf{x}) \cdot \dot{\mathbf{u}} = \mathbf{B} \cdot \dot{\mathbf{u}}$$

- Matrix $\mathbf{B} \in \mathbf{R}^{6 \times 12}$ is constant (independent of x, y, z)
- \mathbf{B} depends on the original geometry of the tetrahedron only

Stress and Energy

Stress as a function of the displacements:

$$\boldsymbol{\sigma} = \mathbf{E} \boldsymbol{\varepsilon} = \mathbf{E} \mathbf{B} \cdot \dot{\mathbf{u}}$$

Energy as a function of the displacements:

$$\begin{aligned} U_{\text{element}} &= \int_{\text{element}} \left(\frac{1}{2} \boldsymbol{\varepsilon}^T \mathbf{E} \boldsymbol{\varepsilon} \right) dV \\ &= \int_{\text{element}} \frac{1}{2} \dot{\mathbf{u}}^T \mathbf{B}^T \mathbf{E} \mathbf{B} \dot{\mathbf{u}} dV \\ &= \frac{1}{2} \dot{\mathbf{u}}^T \left[\int_{\text{element}} \mathbf{B}^T \mathbf{E} \mathbf{B} dV \right] \dot{\mathbf{u}} = \frac{1}{2} \dot{\mathbf{u}}^T \left[V_{\text{element}} \mathbf{B}^T \mathbf{E} \mathbf{B} \right] \dot{\mathbf{u}} \\ &= \frac{1}{2} \dot{\mathbf{u}}^T \mathbf{K} \dot{\mathbf{u}} \end{aligned}$$

Stiffness Matrix

$$U_{\text{body}} = \frac{1}{2} \dot{\mathbf{u}}^T \mathbf{K} \dot{\mathbf{u}}$$

Forces are the derivatives of the energy with respect to the degrees of freedom:

$$\frac{\partial U_{\text{body}}}{\partial \dot{\mathbf{u}}} = \mathbf{K} \dot{\mathbf{u}} = \mathbf{f}$$

The matrix $\mathbf{K} \in \mathbf{R}^{12 \times 12}$ is the stiffness matrix of the element!

$$\mathbf{K} = \int_{\text{element}} \mathbf{B}^T \mathbf{E} \mathbf{B} dV = V_{\text{element}} \mathbf{B}^T \mathbf{E} \mathbf{B}$$

Assembly of elements

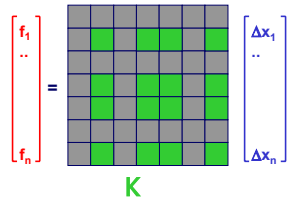
Single element:

$$\begin{bmatrix} f_{1x} \\ f_{1y} \\ f_{1z} \\ \dots \\ f_{4x} \\ f_{4y} \\ f_{4z} \end{bmatrix} = \begin{matrix} \mathbf{K}_e \\ \text{3x3} \end{matrix} \begin{bmatrix} \Delta x_{1x} \\ \Delta x_{1y} \\ \Delta x_{1z} \\ \dots \\ \Delta x_{4x} \\ \Delta x_{4y} \\ \Delta x_{4z} \end{bmatrix}$$

Entire body:

$$\begin{bmatrix} f_1 \\ \dots \\ f_n \end{bmatrix} = \begin{matrix} \mathbf{K} \\ \text{grid} \end{matrix} \begin{bmatrix} \Delta x_1 \\ \dots \\ \Delta x_n \end{bmatrix}$$

Implementation



- K is sparse
- 3×3 block at $(3i, 3j)$ describes how Δx_j influences f_i
- every vertex stores adjacency list of $(3 \times 3$ -matrix, vertex-reference) pairs

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Plastic Strain

An element is under strain ϵ_0 due to displacements \hat{u} :

$$\epsilon_0 = [\epsilon_x, \epsilon_y, \epsilon_z, \gamma_{xy}, \gamma_{yz}, \gamma_{zx}]^T = B \cdot \hat{u}$$

A plastic element „stores“ strain in a state variable:

$$\epsilon_{\text{plastic}}$$

The elastic strain (that causes internal forces) is now:

$$\epsilon_{\text{elastic}} = \epsilon_0 - \epsilon_{\text{plastic}}$$

→ No internal forces are present when $\epsilon_0 = \epsilon_{\text{plastic}}$

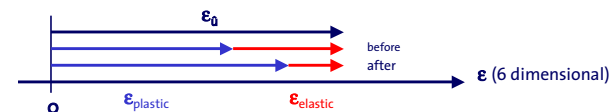
Might be for $\hat{u} \neq 0!$

Plastic Update Rules

Initialization $\epsilon_{\text{plastic}} = 0$

Update rule (every time step):

- Compute $\epsilon_0 = B \cdot \hat{u}$ from actual displacements
- Compute $\epsilon_{\text{elastic}} = \epsilon_0 - \epsilon_{\text{plastic}}$
- if $\|\epsilon_{\text{elastic}}\| > \text{yield}$ then $\epsilon_{\text{plastic}} = \epsilon_{\text{plastic}} + \text{creep} \cdot \epsilon_{\text{elastic}}$
- if $\|\epsilon_{\text{plastic}}\| > \text{max}$ then $\epsilon_{\text{plastic}} = \epsilon_{\text{plastic}} \cdot \text{max} / \|\epsilon_{\text{plastic}}\|$



Implementation

Since

$$\boldsymbol{\varepsilon} = \mathbf{B} \cdot \hat{\mathbf{u}}$$

the displacements that correspond to the plastic strain are:

$$\hat{\mathbf{u}}_{\text{plastic}} = \mathbf{B}^{-1} \cdot \boldsymbol{\varepsilon}_{\text{plastic}}$$

and the corresponding forces are:

$$\mathbf{f}_{\text{plastic}} = \mathbf{K} \mathbf{B}^{-1} \cdot \boldsymbol{\varepsilon}_{\text{plastic}} = [\mathbf{V} \mathbf{B}^T \mathbf{E} \mathbf{B}] \mathbf{B}^{-1} \cdot \boldsymbol{\varepsilon}_{\text{plastic}} = \mathbf{V} \mathbf{B}^T \mathbf{E} \cdot \boldsymbol{\varepsilon}_{\text{plastic}}$$

→ add plastic forces to **external forces**

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Fracture criterion

- Break if internal elastic force exceeds threshold
- stress is a tensor
- the force w.r.t. normal \mathbf{n}_A is:

$$\frac{d\mathbf{f}}{dA} = \boldsymbol{\sigma} \cdot \mathbf{n}_A = \begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{xy} & \sigma_y & \tau_{yz} \\ \tau_{xz} & \tau_{yz} & \sigma_z \end{bmatrix} \cdot \mathbf{n}_A$$

- Find \mathbf{n}_{max} such that $d\mathbf{f}/dA$ is maximal!
- \mathbf{n}_{max} is direction of maximal tensile stress

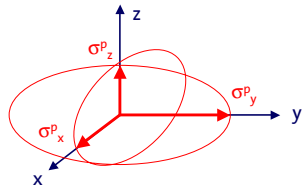
Principal Stresses

- The stress tensor $\boldsymbol{\sigma}$ is symmetric
- → there is a rotation matrix \mathbf{R} such that

$$\begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{xy} & \sigma_y & \tau_{yz} \\ \tau_{xz} & \tau_{yz} & \sigma_z \end{bmatrix} = \mathbf{R}^T \begin{bmatrix} \sigma_x^p & 0 & 0 \\ 0 & \sigma_y^p & 0 \\ 0 & 0 & \sigma_z^p \end{bmatrix} \mathbf{R}$$

- the diagonal entries are the eigenvalues of $\boldsymbol{\sigma}$
- the columns of \mathbf{R} are the corresponding eigenvectors
- there is always a rotated coordinate system where $\boldsymbol{\sigma}$ is diagonal!

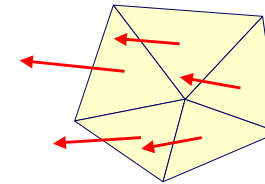
Principal Stresses



- the σ^p are the principal (extremal) stresses
- \rightarrow find maximal eigenvalue of σ
- corresponding eigenvector is the direction of maximal stress

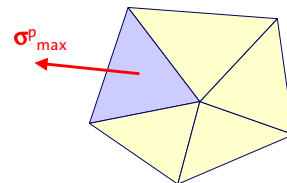
Crack computation

- for all elements: compute maximal tensile stress σ^p_{max}



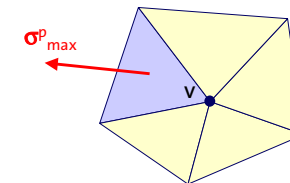
Crack computation

- for all elements: compute maximal tensile stress σ^p_{max}
- if σ^p_{max} exceeds yield stress



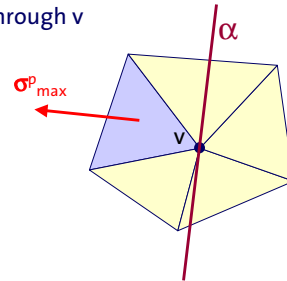
Crack computation

- for all elements: compute maximal tensile stress σ^p_{max}
- if σ^p_{max} exceeds yield stress
- select a vertex v (crack tip / random)



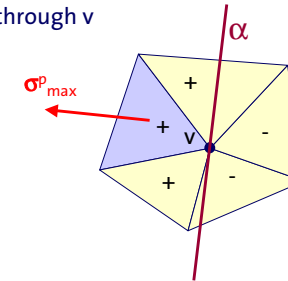
Crack computation

- for all elements: compute maximal tensile stress σ_{max}^p
- if σ_{max}^p exceeds yield stress
- select a vertex v (crack tip / random)
- set plane α normal σ_{max}^p to through v



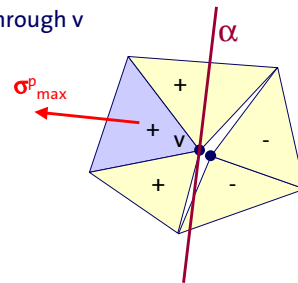
Crack computation

- for all elements: compute maximal tensile stress σ_{max}^p
- if σ_{max}^p exceeds yield stress
- select a vertex v (crack tip / random)
- set plane α normal σ_{max}^p to through v
- mark tetras w.r.t. α



Crack computation

- for all elements: compute maximal tensile stress σ_{max}^p
- if σ_{max}^p exceeds yield stress
- select a vertex v (crack tip / random)
- set plane α normal σ_{max}^p to through v
- mark tetras w.r.t. α
- split vertex



The End

Thank you for your attention!

